#### United States of America

### Department of Transportation -- Federal Aviation Administration

## Supplemental Type Certificate

Number SA01535WI-D

This certificate issued to

Garmin International, Inc. 1200 East 151<sup>st</sup> Street Olathe, KS 66062

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the airworthiness requirements of Part 23\* of the Federal Aviation Regulations.

Original Product-Type Certificate

Number:

A24CE

Make:

Hawker Beechcraft

Model

200, 200C, B200, B200C

300, 300LW, B300, B300C

Description of Type Design Change:

Installation of the Garmin G1000 Integrated Avionics System and GFC 700 Automatic Flight Control System and optional Electronic Stability and Protection (ESP)

### Required Data for Model 200 series:

- (1) Garmin Master Drawing List (MDL) 005-00421-00, Revision 3, dated 3/14/2009
- (2) Garmin Airplane Flight Manual Supplement (AFMS) 190-00915-02, Revision 1, dated 3/14/2009

### Required Data for Model 300 series:

- (1) Garmin Master Drawing List (MDL) 005-00629-00, Revision 1, dated 5/11/2012
- (2) 300, 300LW Garmin Airplane Flight Manual Supplement (AFMS) 190-00716-02, Revision 1, dated 5/11/2012
- (3) B300, B300C Garmin Airplane Flight Manual Supplement (AFMS) 190-00716-03, Revision 1, dated 5/11/2012

Later FAA-approved revisions to the data listed above are incorporated without further revision to this Supplemental Type Certificate.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration

Date of application February 29, 2008

Date of issuance: March 14, 2009

Date reissued :

Date amended: December 14, 2009, November 23, 2010 May 11, 2012, May 14, 2013

By direction of the Administrator

(Signature)

Robert Murray

ODA STC Unit Administrator

ODA-240087-CE

Garmin International, Inc.

(Title)



INSTRUCTIONS: The transfer endorsement below may be used to notify the appropriate FAA Regional Office of the transfer of this Supplemental Type Certificate. The FAA will reissue the certificate in the name of the transferee and forward it to him. Transfer the ownership of Supplement Type Certificate Number To (Name of transferee) Print or type (Address of granter) (Number and street) (City, State and Zip code) From (Name of granter) Print or type (Address of granter)\_\_\_ (Number and street) (City, State and Zip code) Extent of Authority (if licensing agreement): Date of Transfer: Signature of grantor /Tn Tnk/:\_\_\_\_\_

#### United States of America

### Department of Transportation -- Federal Aviation Administration

# Supplemental Type Certificate

(Continuation Sheet)

Number SA01535WI-D

Date of issuance: March 14, 2009

December 14, 2009; November 23, 2010; May 11, 2012;

May 14, 2013

### Limitations and Conditions.

- (1) Compatibility of this design change with previously approved modifications must be determined by the installer.
- (2) Additional airframe, engine and propeller configuration requirements are defined in the data shown on the Garmin Master Drawing List (MDL) identified above.
- (3) Aircraft modified by this STC must be operated in accordance with the Airplane Flight Manual Supplement (AFMS) identified
- (4) Aircraft modified by this STC must be maintained in accordance with the Instructions for Continued Airworthiness (ICA) listed in the Master Drawing List (MDL) identified above.
- (5) If the holder agrees to permit another person to use this certificate to alter the product, the holder must give the other person written evidence of that permission.
- (6) This STC provides Group Airworthiness Approval for Reduced Vertical Separation Minimum (RVSM).
- (7) The Model 200 series optional Electronic Stability and Protection requires MDL 005-00421-00 Rev 9 or later and AFMS 190-00915-02 Rev 4 or later.
- (8) Aircraft, s/n FL-1 to FL-110, FM-1 to FM-8 and FN-1, are approved for improved field performance when modified by Beech Kit 130-3004 prior to or concurrent with this STC installation. The Hawker Beechcraft Airplane Flight Manual Supplement (AFMS) p/n 130-590031-75, must be attached to the POH/AFM for improved performance operation.

### Department of Transportation -- Federal Aviation Administration

# Supplemental Type Certificate

## (Continuation Sheet)

Number SA01535WI-D

Date of issuance: March 14, 2009

Date Amended: December 14, 2009; November 23, 2010; May 11, 2012;

May 14, 2013

#### \*Certification Basis

Based on 14 CFR §§ 21.115 and 21.101, and the FAA policy for significant changes in FAA Order 8110.48, the certification basis for this change is as follows:

- a. The certification basis for parts not changed or not affected by this change is shown on TCDS A24CE.
- b. The certification basis for parts changed or affected by this change is:

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Section	Amdt	Section	Amdt	Section	<u>Amdt</u>
23.143	23-50	23.677 a,d	23-49	23.1329 a,b,c,d,e,f,g,h	23-49
23.181 b	23-45	23.681 a	Original	23.1331 a, b(1)	23-43
23.207 a, b	23-7	23.683	23-7	23.1335	23-20
23.251	23-45	23.685 a,c,d	23-17	23.1351 a,g	23-49
23.301 a	23-48	23.689 a,b,c,d,e	23-7	23.1353 h	23-49
23.303	Original	23.771 a	23-14	23.1357 a,b,c,d	23-43
23.305	23-45	23.773 a(2)	23-45	23.1359	23-49
23.307 a	Original	23.777 a,b	23-51	23.1365 a,b,d,e	23-49
23.337	23-48	23.779 a	23-51	23.1367	Original
23.365 d	Original	23.853 a	23-49	23.1381 a,b	Original
23.395 a(1)	23-7	23.863	23-34	23.1419 a,c	23-43
23.397	23-45	23.867 a,b,c	23-49	23.1431 a,b,e	23-49
23.561 e	23-34	23.901 e	23-53	23.1501	23-21
23.571 c	23-48	23.1301	23-20	23.1521 a,b,c	23-50
23.573 b	23-48	23.1303 a,b,d,f	23-49	23.1523	23-34
23.601	Original	23.1305 a(2)(3),c(1)(2)(5),e(1)	23-52	23.1525	23-45
23.603	23-23	23.1307	23-49	23.1529	23-26
23.605 a	23-23	23.1308 a,b,c	23-57	23.1541 a,b	23-21
23.607	23-48	23.1309 a(1)(3),b,c,d,e,f	23-49	23.1543 b,c	23-50
23.609	Original	23.1311 a,b,c	23-49	23.1545 a,b,d	23-50
23.611	23-48	23.1321	23-49	23.1549	23-45
23.613 a,b,c	23-45	23.1322	23-43	23.1555 a,b,e(2)	23-50
23.627	Original	23.1323 a,b,c	23-49	23.1581 a,b(1)(3),c,d,f	23-50
23.671	Original	23.1325 a,b,e	23-50	23.1583 e,h,m	23-50
	-	23.1327	23-20	23.1585 a(1),j	23-50

Additional requirements for the Models 300 and 300LW:

Section	<u>Amdt</u>
23.149 a,b,c,d,e	23-50
23.672 b, c	23-45
23 1303 e	23-49

Additional requirements for the Models B300 and B300C:

Section	<u>Amdt</u>	Section	<u>Amdt</u>			
23.149 a,b,c,d,e,f	23-50	23.1309 a(4)	23-49			
23.672 b, c	23-45	23.1323 e	23-49			
23.1303 e,g	23-49					
END						