

PURPOSE: (1) TO SYNCHRONIZE THE LCC CLOCK WITH THE CMC CLOCK.

ASSUMPTIONS: (1) SYNCHRONIZATION REQUIRES THE VALUE OF THE DIFFERENCE BETWEEN THE TWO CLOCKS. THIS VALUE MAY BE OBTAINED DIRECTLY FROM CO-ORDINATED READING OF THE TWO CLOCKS OR MAY BE PROVIDED BY THE GROUND FROM DOWNLINK MONITORING OF THE CLOCKS.

(2) THE CMC IS ON. THE LGC MUST ONLY BE ON UNTIL SATISFACTORY COMPARATIVE READINGS HAVE BEEN MADE BY THE CREW OR GROUND.

(3) THIS ROUTINE IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

[illegible]

HOLD
.....
SNAP

CONTINUE UPON
RECEIPT OF ENTER

RECORD CONTENTS OF
CMC CLOCK AT TIME
OF ENTER

DISPLAY CMC CLOCK
TIME ON DSKY:
V06 N65
R1-HRS
R2-MIN
R3-SECS

R3 TO NEAREST .01
SEC.

PERFORM COUNT-
DOWN WITH CREW
MEMBER IN
OTHER VEHICLE
TO INSURE
SIMULTANEOUS
DEPRESSION OF
BOTH ENTER
KEYS.

KEY IN ENTER

MONITOR DSKY:
OBSERVE DIS-
PLAY OF CMC
CLOCK AT TIME
OF ENTER

RECORD THIS
TIME

#40

#50

#60

#70

#80

 WAIT FOR KEYBOARD
 ENTRY

 OBTAIN FROM
 THE LGC THE
 LGC CLOCK TIME
 OF SIMULTAN-
 EOUS ENTER IN-
 TO THE LGC
 (VIA VOICE
 LINK).

 COMPUTE TIME
 DIFFERENCE BE-
 TWEEN COMPU-
 TER CLOCK
 TIMES.

 DO I WISH TO
 TAKE MORE DATA
 POINTS?

.Y .N

 OBTAIN AVER-
 AGE OF ALL
 TIME DIFFER-
 ENCES AND
 TRANSMIT TO
 LM CREW.

 TERMINATE DISPLAY
 UPON RECEIPT OF KEY
 RELEASE

 PUSH KEY RELEASE
 BUTTON

R33/COLOSSUS
 R33/SUNDANCE
 R33/LUMINARY

#90

#100

#110

#120

#130

#140

R33/COLOSSUS
 R33/SUNDANCE
 R33/LUMINARY

.
. .
. .
. .
. .
EXIT

.
. .
. .
. .
. .
EXIT

R33/COLOSSUS
R33/SUNDANCE
R33/LUMINARY

CHANGE CONTROL NOTES

LOGIC REV 06 PCR MIT 66

RENDEZVOUS PARAMETER DISPLAY ROUTINE NO 2 (R34)

LOGIC REV 07 11/27/68

PURPOSE: (1) TO DISPLAY AT ASTRONAUT REQUEST CMC CALCULATED RENDEZVOUS PARAMETERS (RANGE, RANGE RATE, PHI)

ASSUMPTIONS: (1) RANGE AND RANGE RATE ARE CALCULATED BY THE CMC ON THE BASIS OF THE STOPPED LM AND CSM STATE VECTORS AND DO NOT REQUIRE THAT THE ISS BE ON. THE ISS MUST BE ON AND ALIGNED TO A "KNOWN" ORIENTATION AND THE OPTICAL SUBSYSTEM MUST BE ON AND OPERATIONAL, IF A CORRECT DISPLAY OF PHI IS DESIRED. THE RANGE/RANGE RATE/PHI DISPLAY IS NOT INHIBITED HOWEVER IF THE IMU IS NOT ON AND ALIGNED.

(2) THE ROUTINE IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

PRG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
			.CREW .ROUTINE .SELECTION ...			
	START RENDEZVOUS PARAMETER DISPLAY ROUTINE NO 2 (R34)		KEY IN VR5E			#10
	.		.			
	.		.			
++	.		.			
+07	IS ANOTHER EXTENDED VERR ACTIVE?		.			#20
+			.			
+	N.	V.	.			
+	.	.	.			
+	.	.	.			
+	TURN ON OPERATOR ERROR LIGHT		.			
+			.			
+	.	.	.			
+	.	.	.			#30
+07	.	.	.			
++	...		.			
PCN	.	.	.			
586	EXIT		.			
	.		.			
	.		.			
	.		.			

SET TE=PRESENT TIME.

EXTRAPOLATE CSM AND
LM STATE VECTORS TO
T USING COASTING
INTEGRATION ROUTINE.

IS AVERAGE G
RUNNING?

N

SET TE=PRESENT
TIME.

EXTRAPOLATE LM
AND CSM STATE
VECTORS TO TE FROM
T USING KEPLER
SUBROUTINE.

SET TE=AVERAGE G
TIME.

#40

#50

#60

#70

#80

++
+07
++
EDIT

EXTRAPOLATE LM
STATE VECTOR TO
TE FROM T USING
COASTING INTEG-
RATION ROUTINE.

#90

CALCULATE RANGE,
RANGE RATE AND
PHI

#100

HOLD

MON

FLASH VERR-NOUN TO
REQUEST RESPONSE AND
DISPLAY RENDEZVOUS
PARAMETERS:
VIA N53
R1-RANGE
R2-RANGE RATE
R3-PHI

MONITOR DSKY:
OBSERVE VERR-NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
OF RENDEZVOUS
PARAMETERS.
(NOTE: THESE PARA-
METERS WILL BE
UPDATED EVERY TWO
SECONDS.

#110

RANGE-CALCULATED
RANGE TO LM. IN
NAUTICAL MILES TO
NEAREST .01 NM.

#120

RANGE RATE-
CALCULATED RANGE
RATE BETWEEN CSM AND
LM. NEGATIVE SIGN
INDICATES CLOSING
IN EPS TO NEAREST
.1 EPS
PHI-ANGLE BETWEEN
OPTICS STAR LINE OF
SIGHT AND THE LOCAL
HORIZONTAL PLANE AT
THE PRESENT TIME.

#130

ANGLE IS ALWAYS
POSITIVE.
FROM 0 TO 360

++
+07
++

#140

EDIT . DEGREES IN DEGREES
 . TO NEAREST .01
 . DEGREE

++
+06
++
PCR
511

WAIT FOR KEYBOARD
ENTRY

TERMINATE FLASH UPON
RECEIPT OF PROCEED

• PROCEED

EXIT 024

WHEN FINISHED WITH
THIS DISPLAY KEY IN
PROCEED

EXIT R34

#150

#160

#179

CHANGE CONTROL NOTES

LOGIC REV 04 PCP MJT 66
REV 05 PCP 206
PCP 495
REV 06 PCP 512
PCP 511
REV 07 PCP 586

LUNAR LANDMARK SELECTION ROUTINE (R35)

LOGIC REV 09 11/27/68

PURPOSE: (1) TO DISPLAY TO THE ASTRONAUT THE TIME AT WHICH THE SPACECRAFT WILL PASS OVER THE LANDING SITE.
 (2) TO CALCULATE AND DISPLAY FIVE LANDMARKS WHICH WILL BE SUITABLE FOR NAVIGATION.
 (3) TO CALCULATE AND DISPLAY THE TIME AT WHICH THE SPACECRAFT WILL PASS OVER EACH LANDMARK.

ASSUMPTIONS: (1) THIS ROUTINE SHOULD BE SELECTED BY THE NAVIGATOR IN LUNAR ORBIT PRIOR TO SELECTION OF THE ORBITAL NAVIGATION PROGRAM (P22).
 (2) THIS ROUTINE MAY BE SELECTED ONLY DURING THE CMC IDLING PROGRAM (P00).
 (3) THE SELECTED LANDMARKS ARE NOT NECESSARILY DISPLAYED IN CHRONOLOGICAL ORDER.

++
 +09
 ++
 EDIT

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
--------------	-----	--------	------	-----------	------	---------------

.CREW ROUTINE
 .SELECTION

...

 .
 START LUNAR LMK SEL-
 ECTION ROUTINE

 KEY IN V79E

.
 .
 .

#10

 IS CURRENT PROGRAM
 P00?

.Y	.N
.	.
.	.

 IS ANOTHER EX-
 TENDED VERB
 ACTIVE?

#20

N.	.Y
.	.
.	.
.	.
.	.
.	.
.	.
.	.
.	.
.	.
.	.

++
 +09
 ++
 PCN
 586

EXIT

•Y •N

IN ORDER TO
USE THIS
ROUTINE KEY
IN V37E00E
AND THEN KEY
IN V79E

EXIT

```
FLASH VERB-NCUN TO
REQFST RESPONSE AND
DISPLAY T-LAT-LONG
V06N34
R1 00XXX
R2 000XX
R3 0XXXX
IN HOURS-MIN-SEC TO
THE NEAREST .01 SEC
NOTE: FIRST
DISPLAY WILL RE
PRESENT G.E.T.
```

T-LAT-LONG=C.E.T.
THIS IS THE TIME
WHICH THE
COMPUTER WILL DE-
FINE AS THE START OF
LMK TRACKING. IT
WILL DETERMINE THE
NEXT TIME THE LAND-
ING SITE WILL PASS
UNDER THE SPACECRAFT
AND IT WILL DETER-
MINE 5 LANDMARKS
WHICH WILL PASS
UNDER THE SPACECRAFT
AFTER T-LAT-LONG AND
WHICH WILL BE ALONG
THE GROUND TRACK.

80

WAIT FOR KEYBOARD
ENTRY

MONITOR DSKY:
OBSERVE VERB NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
THE NEXT TIME AFTER
T-LAT-LONG AT WHICH
THE LANDING SITE
WILL PASS UNDER THE
SPACECRAFT.

RECORD THIS TIME. DO
I WISH TO CONTINUE
WITH THIS ROUTINE
TO OBTAIN LANDMARK
INFORMATION?

.N	.Y
.	.
.	.
.	.

KEY IN
TERM-
NATE,
V34E

KEY IN
PROCEED

TERMINATE FLASH UPON
RECEIPT OF PROCEED
OR TERMINATE

EXIT R35

```

.P          .T
.R          .E
.O          .R
.C          .M
.E          .I
.E          .N
.D          .A
.           .T
.           .E
.           .
.           .
.           .
.           .
EXIT R35

```

180

#200

#210

#220

230

#240

R2 LMK ID NO

WAIT FOR KEYBOARD
ENTRY

NOTE: THE COMPUTER
WILL CALCULATE 5
LMKS SUITABLE FOR
NAVIGATION AND WILL
DISPLAY THEN
SEQUENTIALLY ON
COMMAND STARTING
WITH THE LANDMARK
ASSOCIATED WITH THE
+60 DEGREE LONGITUDE
LINE AND CONTINUING
THROUGH TO THE LAND-
MARK ASSOCIATED WITH
THE -60 DEGREE
LONGITUDE LINE.

RECORD DATA. DO I
WISH DISPLAY OF THE
TIME AT WHICH THIS
LNK WILL PASS UNDER
THE SPACECRAFT?

$\bullet$.N $\bullet$.Y

KEY IN PROCEED

HAVE ALL 5 LANDMARKS
BEEN DISPLAYED?

N **Y**

536

DO I WISH TO
SEE THE LAND-
MARK CODE FOR
THE NEXT LAND-
MARK ON MY
GROUND TRACK?

.Y

.N

KEY IN
TERM-
INATE
V34E

#300

#310

EXIT
R35

#320

TERMINATE FLASH UPON
RECEIPT OF RECYCLE,
TERMINATE OR PROCEED

KEY IN RECYCLE
V32E

.R	.P	.T
.E	.R	.F
.C	.C	.R
.Y	.C	.M
.C	.E	.I
.L	.E	.N
.E	.D	.A
		.T
		.F
		.
		.
		.
		...
		.
		EXIT
		R35

#330

#340

EXIT
R35

#390

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RENDEZVOUS OUT OF PLANE DISPLAY ROUTINE (R36)

LOGIC REV 06 11/27/68

- PURPOSE: (1) TO DISPLAY AT ASTRONAUT REQUEST CMC CALCULATED RENDEZVOUS OUT OF PLANE PARAMETERS (Y,YDOT,PSI)
- ASSUMPTIONS: (1) THESE PARAMETERS ARE CALCULATED BY THE CMC ON THE BASIS OF THE STORED LM AND CSM STATE VECTORS AND DO NOT REQUIRE THAT THE ISS BE ON.
- (2) THE ROUTINE IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
			.CREW .ROUTINE .SELECTION . .			
	START THE RENDEZVOUS OUT OF PLANE ROUTINE		KEY IN V90E			#10
	.					
	.					
++	IS ANOTHER EXTENDED					
+06	VERB ACTIVE?					
+						
+	.N	Y.				
+	.	.				
+	.	.				#20
+	.					
+	. TURN ON					
+	. OPERATOR					
+06	. ERROR LIGHT					
++	.					
PCN	.	.				
586	.	.				
	.	.				
	...					
	.					#30
	. EXIT					
	.					
	.					
						
	.	.				
	.	.				
	.	.				
	.	.				
	.	.				

R36/COLOSSUS
R36/SUNDANCE
R36/LUMINARY

#40

HOLD
.....
SNAP
++
+06
++
PCR
206
FDIT

FLASH VERB NOUN TO
REQUEST RESPONSE AND
DISPLAY T(EVENT) IN
G.E.T.:
V06 N16
R1-T(EVENT)-HRS
R2-T(EVENT)-MINS
R3-T(EVENT)-SECS

T(EVENT): TIME
(G.E.T.) FOR WHICH
OUT OF PLANE PARA-
METERS ARE DESIRED
IN HRS, MINS, AND
SECS TO NEAREST .01
SECONDS.
A SPECIAL CASE IS
ALL ZERCS INDICA-
TING PRESENT TIME

MONITOR DSKY:
OBSERVE VERB NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
OF TIME AT WHICH
OUT OF PLANE PARA-
METERS ARE DESIRED.

DO I WISH TO HAVE
THE CMC COMPUTE
PARAMETERS FOR THE
PRESENT TIME?

.N .Y
. . .
. . .
. . .

AM I SATISFIED
WITH THE DIS-
PLAYED TIME?

.N .Y
. . .
. . .
. . .

ARE ALL
THREE
REGISTERS
EQUAL TO
ZERO?

.Y .N
. . .
. . .
. . .

#50

#60

#70

#80

543

++
+06
++
EDIT

R36/COLOSSUS
R36/SUNDANCE
R36/LUMINARY

#90

#100

#110

#120

#130

R36/COLOSSUS
R36/SUNDANCE
R36/LUMINARY

WAIT FOR KEYBOARD
ENTRY: .

KEY IN
PROCEED

TERMINATE FLASH UPON
RECEIPT OF PROCEED
OR NEW DATA. .

KEY IN V25E AND LOAD
NEW DATA.

NEW .P
DATA .R
C
C
STORE NEW .E
DATA .E
D

IS T(EVENT) ZERO?

N .Y
.

EXTRAPOLATE CSM
AND LM VECTORS
TO THE PRESENT
TIME USING
PRECISION
INTEGRATION

EXTRAPOLATE CSM
AND LM STATE
VECTORS TO THE
TIME DEFINED BY
T(EVENT) USING
PRECISION
INTEGRATION

CALCULATE OUT-OF-
PLANE PARAMETERS:

Y
Y DOT
PSI

#140

HOLD
.....
SNAP

FLASH VERB NOUN TO
REQUEST RESPONSE AND
DISPLAY RENDEZVOUS
OUT OF PLANE PARA-
METERS:

V06 N90
R1 Y
R2 Y DOT
R3 PSI

Y-(NOTE: FOR
DEFINITION OF
PARAMETERS REFER
TO SECTION
5.6.7.4 OF THIS
DOCUMENT.) IN
NAUTICAL MILES TO
THE NEAREST .01
NM.

Y DOT-RATE OF
CHANGE OF Y (+ IS
INCREASING AND -
IS DECREASING) IN
FPS TO THE NEAR-
EST .1 FPS.

PSI-ANGLE BETWEEN
THE CSM ORBITAL
PLANE AND THE LCS
TO THE LM PRO-
JECTED INTO THE
HORIZONTAL PLANE.
IN DEGREES TO THE
NEAREST .01 DEG-
REES.

MONITOR DSKY:
OBSERVE VERB NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
OF RENDEZVOUS OUT OF
PLANE PARAMETERS

#150

DO I WISH TO RECEIVE
ANOTHER DATA POINT
FOR A DIFFERENT
TIME?

#160

.N .Y

#170

#180

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SPS THRUST FAIL ROUTINE (R40)

LOGIC REV 02 07/11/68

PURPOSE: (1) TO INDICATE TO THE ASTRONAUT THAT THE GNCS HAS DETECTED A THRUST FAILURE.

(2) TO PROVIDE THE ASTRONAUT A FLASHING DISPLAY TO WHICH HE CAN RESPOND AS DESCRIBED IN THE FLOW.

ASSUMPTION: (1) THE GNCS HAS DETECTED A THRUST FAILURE AND HAS SHUT OFF CROSS PRODUCT STEERING AND HAS STOPPED THE CALCULATIONS OF TIME FROM CUTOFF AND HAS STOPPED C.G. TRACKING.

(2) IF THE ASTRONAUT KEYS IN PROCEED ON THIS DISPLAY THRUST FAILURE DETECTION WILL BE INHIBITED FOR 2 SECONDS TO PREVENT A PREMATURE THRUST FAIL INDICATION.

(3) THIS ROUTINE IS SELECTED BY THE SPS THRUSTING PROGRAM (P40)

[illegible]

WAIT FOR KEYBOARD
ENTRY

SHALL I TERMINATE
THE ENGINE ON COMM-
AND AND RETURN TO
THE V99 FLASH WHICH
WILL ALLOW ME TO
EITHER REISSUE THE
ENGINE ON COMMAND OR
CONTINUE ON TO THE
DISPLAY OF VGX, VGY,
VGZ FOR RCS
THRUSTING?

.Y .N

KEY IN ENTER

HAS THRUST COME
BACK ON?

.Y .N

KEY IN PROCEED

SHALL I TERMINATE
THE ENGINE ON COM-
MAND AND THE SPS
THRUSTING PROGRAM
(P40)?

.Y .N

#30

#40

#50

#60

#70

KEY IN TERMINATE
V34F

COMMAND
ENGINE
OFF

GO TO
"B"
IN P40

DD
ROUTINE
R00

EXIT
R40 AND
P40

TERMINATE THE
ENGINE ON
COMMAND

WAIT ABOUT
2.5 SEC

DO
ROUTINE
R00

EXIT
R40 AND
P40

#80

#90

100

#110

120

TURN OFF
TVC DAP

SET NARROW
DEADBAND
IN RCS DAP

DRIVE SPS ENGINE
BELL TO TRIM PO-
SITION.
NOTE: THE TRIM
POSITION IS THAT
LAST DEFINED BY
THE C.G. TRACKING
COMPUTATION.

TURN ON RCS
DAP IN .6
SEC

GO TO
"A"
IN P40

#130

#140

#150

#160

CHANGE CONTROL NOTES

REV 01 PCR MIT 98
REV 02 PCR 206

PURPOSE: (1) TO INTEGRATE THE STATE VECTOR OF THIS VEHICLE TO THE TIME AT WHICH THE AVERAGE G ROUTINE WILL BE TURNED ON BY THE CALLING PROGRAM.

(2) TO DEFINE A NEW TIC FOR PROGRAMS 40 OR 41 IN THE EVENT THE STATE VECTOR CAN NOT BE INTEGRATED TO THE TIME DEFINED BY PROGRAMS 40 OR 41 AND TO LIGHT THE ALARM LIGHT TO INFORM THE CREW THAT TIC HAS BEEN SLIPPED.

ASSUMPTIONS: (1) THERE IS A SIGNIFICANT AMOUNT OF TIME REQUIRED BY THE CMC TO TURN ON THE AVERAGE G ROUTINE. THIS TIME IS VARIABLE ACCORDING TO THE FOLLOWING CONSIDERATIONS REGARDING THE CSM STATE VECTOR:

1.4 SECONDS PER TIME STEP IN EARTH ORBIT

2.5 SECONDS PER TIME STEP IN LUNAR ORBIT

WHERE TIME STEP IS EQUAL TO:

240 SECONDS IN EARTH ORBIT

350 SECONDS IN LUNAR ORBIT

711 THE ROUTINE IS ONLY AUTOMATICALLY SELECTED.

PPNG
CONT

CMC

GRIND

CRFW

CHECKLIST

TIME

TOTAL
TIME

CMC ROUTINE
SELECTION

```
. . . . .
```

.

.

.

.

.

.

.

READ PRESENT TIME,

TP

.

.

.

.

IS THE TIG FLAG SET?

	N		Y
.	.	.	.
.	.	.	.
.	.	.	.

IS TRFC-IO SEC
GREATER THAN OR
EQUAL TO TP?

	Y		N
.	.	.	.
.	.	.	.
.	.	.	.

#10

#20

R41/COLDSSUS
R41/LUMINARY

TURN ON PRO-
GRAM ALARM
LIGHT AND
STORE ALARM
CODE 1703

SET TIG
FLAG

SET TDEC=TP PLUS
10 SECONDS

IS INTEGRATION
FINISHED?

.N .V

INTEGRATE THIS
VEHICLES STATE
VECTOR ONE TIME STEP
OR TO TDEC IF LESS
THAN ONE TIME STEP
USING PRECISION
INTEGRATION.

IS THE TIG
FLAG SET?

.N .V

MONITOR OSKY PROGRAM
ALARM LIGHT DURING
THE PERIOD FROM TIG
-40 SECONDS TO
BLANKING AT TIG-35:
IF LIGHT COMES ON
DURING THIS TIME IT
INDICATES THAT TIG
WILL BE SLIPPED AS
REQUIRED TO GET THE
STATE VECTOR INTEG-
RATED TO A NEW TIG
-30 SECONDS.

IF THIS ALARM
CONDITION OCCURS THE
TEI DISPLAY WILL
CONTINUE TO COUNT
BASED ON THE ORIGI-
AL TIG UNTIL INTE-
GRATION IS COMPLETE
AND A NEW TIG IS
ESTABLISHED. THE
DISPLAY WILL NOT
BLANK HOWEVER UNTIL
THE NEW TIG HAS BEEN
ESTABLISHED AND TEI
IS REVISED.

OBSERVE THAT THE
COMPUTER ACTIVITY
LIGHT GOES OFF WHICH
INDICATES THAT THIS
ROUTINE IS COMPLETE

#30

#40

#50

#60

#70

#80

++
+03

552

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AUTOMATIC OPTICS POSITIONING ROUTINE (R52)

LCGIC REV 09 11/27/68

PURPOSE: (1) TO POINT THE STAR LOS OF THE OPTICS AT A STAR OR LANDMARK DEFINED BY THE PROGRAM OR BY DSKY INPUT (ASTRONAUT).
 PCR
 206 (2) TO POINT THE STAR LOS OF THE OPTICS AT THE LM DURING RENDEZVOUS TRACKING OPERATIONS.

++
 +09 (3) TO DO THE TRACKING ATTITUDE ROUTINE (R61) APPROXIMATELY EVERY 2 SECONDS DURING RENDEZVOUS TRACKING OPERATIONS.
 ++
 EDIT

ASSUMPTIONS- (1) THE ROUTINE IS AUTOMATICALLY SELECTED DURING IMU REALIGN PROGRAM (P52) BY THE RENDEZVOUS NAVIGATION PROGRAM (P20), BY THE ORBITAL NAVIGATION PROGRAM (P22), OR BY THE CISELUNAR NAVIGATION PROGRAM (P23).
 (2) THIS ROUTINE IS SELF PERPETUATING AND IS TERMINATED BY THE SIGHTING MARK ROUTINE (R53) FOR STAR OR LMK AND BY RESETTNG THE TRACK FLAG FOR LM.

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
	.CMC ROUTINE .SELECTION					
	----- START AUTO OPTICS POSITIONING ROUTINE (R 52) -----					#10
++ +09 ++ EDIT						
	----- SET TRUNNION DRIVE FLAG -----					#20
						

*40

#50

#60

#70

#80

R50/COLLOSSUS
R50/SUNDANCE
R50/LUMINARY

516

007

•
•
•
•
•

TERMINATE COARSE
ALIGN MODE IN ISS.
RESUME ATTITUDE
HOLD OF VEHICLE

•
•
•
•
•
•

EXIT R50

R50/COLOSSUS
R50/SUNDANCE
R50/LUMINARY

#90

#100

CHANGE CONTROL NOTES

LOGIC REV 03 PCR 464

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AUTOMATIC OPTICS POSITIONING ROUTINE (R52)

LOGIC REV 09 11/27/68

PURPOSE: (1) TO POINT THE STAR LOS OF THE OPTICS AT A STAR OR LANDMARK DEFINED BY THE PROGRAM OR BY DSKY INPUT (ASTRONAUT).
 PCR
 206 (2) TO POINT THE STAR LOS OF THE OPTICS AT THE LM DURING RENDEZVOUS TRACKING OPERATIONS.

++
 +09 (3) TO DO THE TRACKING ATTITUDE ROUTINE (R61) APPROXIMATELY EVERY 2 SECONDS DURING RENDEZVOUS TRACKING OPERATIONS.
 ++
 EDIT

ASSUMPTIONS- (1) THE ROUTINE IS AUTOMATICALLY SELECTED DURING IMU REALIGN PROGRAM (P52) BY THE RENDEZVOUS NAVIGATION PROGRAM (P20), BY THE ORBITAL NAVIGATION PROGRAM (P22), OR BY THE CISELUNAR NAVIGATION PROGRAM (P23).
 (2) THIS ROUTINE IS SELF PERPETUATING AND IS TERMINATED BY THE SIGHTING MARK ROUTINE (R53) FOR STAR OR LMK AND BY RESETING THE TRACK FLAG FOR LM.

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
	.CMC ROUTINE .SELECTION					
	----- START AUTO OPTICS POSITIONING ROUTINE (R 52) -----					#10
++ +09 ++ EDIT						
	----- SET TRUNNION DRIVE FLAG -----					#20
						

IS THE LM TARGET
FLAG SET ?

.

•

•

•

IS THE TRACK FLAG
SFT?
(SEE P20)

•

•

THE LOGIC FROM THIS
POINT TO "B" BELOW
IS FOR THE LM TARGET
CASE ONLY.

#70

 IS THE PREFERRED
 ATTITUDE FLAG SET?

.NO .YES

#80

 IS UPDATE
 FLAG SET?

.N .Y

#90

 WAIT ABOUT 1.3
 SECONDS

....

#100

 . EXTRAPOLATE CSM AND
 . LM STATE VECTORS TO
 . THE PRESENT TIME
 . +1.3 SECONDS USING
 . CONIC EQUATIONS

++
 +09
 ++
 EDIT

#110

 . READ PRESENT VEHICLE
 . ATTITUDE FROM THE
 . ICDU'S

#120

 . COMPUTE TARGET VEC-
 . TOR FROM CSM TO LM

562

• Y • N

MONITOR PROGRAM
ALARM LIGHT: IF THE
PROGRAM ALARM LIGHT
COMES ON AT THIS
TIME THE ASTRONAUT
SHOULD VERIFY ALARM
(BY KEYING IN
V05N09E) AND EITHER
MANUALLY MANEUVER
THE VEHICLE BACK TO
THE PREFERRED TRACK-
ING ATTITUDE
(DESIRED GIMBAL
ANGLES MAY BE DIS-
PLAYED BY KEYING IN
V16N22E AND DESIRED
OPTICS BY KEYING IN
V16N92F) OR ALLOW
THE GNCS TO AUTOMA-
TICALLY MANEUVER BY
SWITCHING TO CMC
AUTO CONTROL AND/OR
BY KEYING IN V58E.

180

• DRIVE	DRIVE	•
• SHAFT	SHAFT	•
• AND	CDU	•
• TRUNNION	ONLY	•
• CDU'S		•

DO THE TRACKING
ATTITUDE ROUTINE
R61.

•
•
•
•
•
•
•

WAIT 0.5 SEC

•
•
•
•

GC TO
"A"
ABOVE

"B" FROM
ABOVE

•
•
•
•
•
•
•
•
•

566

COMPUTE
TARGET
VECTOR
FROM CSM
TO DES-
IGNATED
LAND-
MARK.

COMPUTE TARGET VECTOR
FROM CSM TO
DESIGNATED
STAR.

CALCULATE THE REQUIRED OPTICS ANGLES TO POINT THE STAR LOS OF THE OPTICS ALONG THE TARGET VECTOR.

IS A TRUNNION ANGLE
REQUIRED TO POINT
THE STAR LOS OF THE
OPTICS AT THE TARGET
GREATER THAN 90 DEG?

• N • Y

#430

POSS
PRIO
HOLD .
.....
SNAP .

FLASH VERB-NOUN
TO REQUEST RESP-
ONSE AND DISPLAY
ALARM CODE:

V05NC9

R1-

R2-

R3-

EXPECTED ALARM
ALARM CODE AT
THIS TIME IS 404

WAIT 2 SECONDS

++
+09
++
EDIT
PCR
206

MONITOR DSKY:
DOES ALARM CODE
DISPLAY INDICATE
THAT THE TARGET IS
NOT WITHIN THE
HEMISPHERE OF
OPTICS VISIBILITY?

.Y N.
:
:

FOR STAR/LANDMARK
SIGHTINGS THERE
ARE TWO OPTIONS:

(A) MANUALLY MA-
NEUVER VEHICLE
UNTIL OPTICS CAN
ACQUIRE THE DES-
IRED TARGET.

(B) TERMINATION
OF THE PRO-
GRAM AND
ROUTINE.

.A B.
:
:
:

MANUALLY
MANEUVER VEH-
ICLE UNTIL IT
IS ESTIMATED
THAT OPTICS
CAN ACQUIRE
THE TARGET.
MONITOR FDI
TO AVOID
GIMBAL LOCK.

WAIT FOR KEY-
BOARD ENTRY.

KEY IN
PROCEED

#440

#450

#460

#470

#480

#490

TERMINATE FLASH
UPON RECEIPT OF
PROCEED, OR
TERMINATE.

KEY IN TERMIN-
ATE.
V34E

.T .P
.E .R
.R .O
.M .C
.I .E
.N .E
.A .D
.T .
.E .

WAIT .5 SEC

DO ROUTINE
R00

DO ROUTINE
R00

EXIT R52
AND CALLING
PROGRAM

EXIT R52
AND CALLING
PROGRAM

IS THE TRUNNION
ANGLE REQUIRED TO
POINT THE STAR LOS
OF THE OPTICS AT THE
TARGET GREATER THAN
50 DEGREES?

.N .Y
. .
. .
. .
. .

#500

#510

#520

#530

#540

++
+09
++
EDIT

 SET TRUNNION
 DRIVE FLAG

#550

 .RESET TRUNNION
 .DRIVE FLAG.

 .TURN ON PROGRAM
 .ALARM LIGHT AND
 .STORE ALARM CODE
 .407

 MONITOR PROGRAM
 ALARM LIGHT: IF THE
 PROGRAM ALARM LIGHT
 COMES ON AT THIS
 TIME THE ASTRONAUT
 SHOULD VERIFY THE
 ALARM BY KEYING
 IN V05N09E AND
 THEN REVIEW THE
 OPTICS ANGLES IF
 THEY ARE BEING
 DISPLAYED (THEY WILL
 BE UNLESS R53 HAS
 BEEN PREVIOUSLY
 SELECTED). IF THEY
 ARE NOT BEING
 DISPLAYED KEY IN
 V16N92F. IF TRUNNION
 (R2) IS GREATER THAN
 49.775 THE ASTRONAUT
 SHOULD MANUALLY
 MANEUVER THE
 VEHICLE.

#560

#570

++
 +09
 ++
 EDIT

#580

 IS THE SIGHTING
 MARK FLAG SET?

.Y .N

#590

5 11

R52/COLOSSUS

```

++
POSS . DISPLAY CN DSKY:
..... VO6N92
MON . R1-SHAFT
+09 . R2-TRUNNION
++ . R3-BLANK
EDIT
PCR . SHAFT-DESIRED
206 . SHAFT ANGLE. IN
. DEGREES TO NEAR-
. EST .01 DEGREE.

. TRUNNION-DESIRED
. TRUNNION ANGLE.
. IN DEGREES TO
. NEAREST .001
. DEGREE.

CHECK OPTICS MODE
DISCRETE. IS THE
OSS IN CMC MODE?

.Y .N
.
.

IS TRUNNION
DRIVE FLAG SET?

.Y .N
.
.

DRIVE DRIVE
SHAFT SHAFT
AND CDU
TRUN- ONLY
NION
CDUS

WAIT 0.5 SEC

```

```

-----
MONITOR DSKY:
. OBSERVE DISPLAY
OF DESIRED OPTICS
ANGLES. NOTE: THIS
DISPLAY WILL NOT
APPEAR IF R53 HAS
ALREADY BEEN CALLED
BY SWITCHING OPTICS
MODE SWITCH TO
MANUAL.
-----

```

#600

#610

#620

#630

#640

R52/COLOSSUS

LOGIC REV 07	PCR # MIT 54
LOGIC REV 08	PCN 503
LOGIC REV 09	PCR 206 EDITORIAL
09	PCN 594

ASSUMPTIONS: (1) SIGHTINGS ARE MADE WITH EITHER SCT OR SXT AS REQUIRED.

(3) WHEN THE CMC ACCEPTS A MARK IT RECORDS AND STORES 5 ANGLES (3 JOINS AND 2 BOOTS) AND THE TIME OF THE MARK.

(4) MAXIMUM UNREFLECTED-MARK THRESHOLDS ARE ESTABLISHED UTILIZING A MARK COUNTER. THIS MARK COUNTER WILL INCREMENT EACH TIME A MARK IS MADE AND DECREMENT EACH TIME A MARK REJECT IS MADE. THE MARK COUNTER WILL BE COMPARED AGAINST A MARK INDEX. THE MAGNITUDE OF THIS INDEX WILL BE SELECTED BY THE PROGRAM (OR ROUTINES) SELECTING THE SIGHTING MARK AS FOLLOWS:

(B) LANDMARK - 5

THE ROUTINE REQUIRES THAT AT LEAST ONE MARK BE TAKEN FOR NORMAL TERMINATION. IF FOR SOME REASON (TARGET NOT VISIBLE, DISTINGUISHABLE, ETC), THE ASTRONAUT GETS INTO THIS ROUTINE AND FEELS NOT TO MARK HE SHOULD EITHER KEY IN V34F WHICH WILL CALL R00 OR CALL A NEW PROGRAM BY KEYING IN V37EXX.

(5) THE ROTATION OR MINIMUM IMPULSE CONTROLLER MAY BE USED AS REQUIRED TO REDUCE THE S/C ORBIT RATE.

(6) THE ROUTINE IS SELECTED IN THE AUTO OPTICS POSITIONING ROUTINE (P52) BY CREW SELECTION OF MANUAL OPTICS MODE.

(7) THE ROUTINE MAY ALSO BE ENTERED DIRECTLY FROM P51 OR 023.

TOTAL
TIME

•

• • •

START SIGHTING MARK
ROUTINE (R 53)

•
•
•
•
•
•
•
•

#20

#30

#40

#50

#60

SET SIGHTING
MARK FLAG

ZERO MARK COUNTER

"A"

"A"

HOLD .
.....
SNAP .

FLASH VERR-
TO REQUEST PLEASE
MARK:
V51 N BLANK
P1-BLANK
P2-BLANK
P3-BLANK

MONITOR DSKY:
OBSERVE FLASHING
VERR TO REQUEST
PLEASE MARK

SELECT MANUAL OPTICS
MODE

WAIT FOR KEYBOARD
ENTRY

WHEN SIGHTING IS
SATISFACTORILY
ACCOMPLISHED PRESS
MARK BUTTON

WAS THE SIGHTING
SATISFACTORY?

Y. N
.
.
.
PRESS MARK
REJECT
BUTTON

#70

GO TO
"A"
ABOVE

IS THE TARGET A
CELESTIAL BODY?

#90

DO I WISH TO
MAKE MORE
MARKS?

#90

GO TO
"A"
ABOVE

TERMINATE FLASH OF
V51 UPON RECEIPT
OF MARK, MARK
REJECT, OR PROCEED

MARK . M. D
REJECT . A. D
D. C
K. C

IS MARK FLAG
SET?

N. V.

KEY IN
PROCEED

#100

#110

#120

GO TO
"A"
ABOVE

ERASE LAST SET
OF MARK DATA AND
DECREMENT MARK
COUNTER BY ONE.

·
·
·

RESET MARK FLAG

·
·
·

GO TO
"A"
ABOVE

·
·
·

SET MARK
FLAG

·
·
·

STORE FIVE ANGLES
AND TIME AND
INCREMENT MARK
COUNTER BY ONE.

·
·
·

IS MARK COUNTER
STILL LESS THAN
MARK INDEX?

·
·
·
·
·
·
·

·
·
·
·
·
·
·

```

ERASE LAST SET
OF MARK DATA AND
DECREMENT MARK
COUNTER BY ONE.

```

RESET MARK FLAG

GO TO
"A"
ARJVF

SET MARK
FLAG

STORE FIVE ANGLES
AND TIME AND
INCREMENT MARK
COUNTER BY ONE.

IS MARK COUNTED
STILL LESS THAN
MARK INDEX?

$\begin{array}{c} \bullet \\ \bullet \\ \bullet \\ \bullet \end{array} \quad \begin{array}{c} \bullet \\ \bullet \\ \bullet \\ \bullet \end{array}$

#170

#180

#190

#200

#210

#220

#230

HOLD...
SNAP...

GO TO
"A"
ABOVE

FLASH VERB MOUN TO
REQUEST PLEASE
PERFORM TERMINATE
MARK SEQUENCE:
V50 N25
P1-00016
P2-BLANK
P3-BLANK

WAIT FOR KEYBOARD
ENTRY.

MONITOR DISKY:
OBSERVE FLASHING
VERB-MOUN TO REQUEST
PLEASE PERFORM
TERMINATE MARK
SEQUENCE

WAS LAST MARK
SATISFACTORY?

Y N

KEY IN
PROCEED

PRESS MARK
REJECT
BUTTON

GO TO
"A"
ABOVE

TERMINATE FLASH
UPON RECEIPT OF
ENTER, OR MARK
REJECT

MARK P
REJ P
C
C
F
F IS MARK
D COUNTER
= 0?
Y N

GO TO
MARK
ABOVE

RESET MARK FLAG

IS THE CALLING
PROGRAM P22 OR P23?

N Y
SET TERMINATE
FLAG (FOR USE
BY P52)

SET TERMINATE
FLAG (FOR USE
BY P52)

++
+00
+
+00
++
PCM
660

#240

#250

#260

#270

#280

#790

4300

4319

#320

4330

HOLP .
.....
SNAP .

FLASH VERB-ADJ TO
REQUEST RESPONSE AND
DISPLAY OF CELESTIAL
BODY CODE

V01 N71
P1-000XX
P2-BLANK
P3-BLANK

CELESTIAL BODY
GIVE THE DESIGNATION
OF THE CELESTIAL
BODY WHICH WAS MARK-
ED (IN OCTAL).

00 - PLANET (ANY
PLANET)
01/45 - STAR (FROM
CELESTIAL BODY
CODE LIST)
46 - SUN
47 - EARTH
50 - MOON

WAIT FOR KEYBOARD
ENTRY

MONITOR DSKY:
OBSERVE FLASHING
VEEB-NOUN TO REQUEST
RESPONSE AND DISPLAY
OF CELESTIAL BODY
CODE.

AM I SATISFIED WITH
THIS CODE?

• V • R

KEY IN
APPROVED

- KEY IN V21F AND
- LOAD CELESTIAL
- BODY CODE:

#389

#300

#400

#410

#420

#430

POSS
HOLD
.....
SMAP

++
+00
++
EDIT

CALCULATE CEL-
ESTIAL BODY
VECTOR FOR THE
BODY DEFINED
BY THE STAR
CODE.

FLASH VERR-
NOUN TO RE-
QUEST RESPONSE
AND DISPLAY
PLANET POSI-
TION VECTOR;
V06 NPP
P1-Y PL
P2-Y PL
P3-Z PL

X PL - THE Y
COMPONENT OF 1/2
UNIT POSITION
VECTOR OF THE
PLANET AT GET.
IN REFERENCE
COORDINATES.
TO THE FIFTH
PLACE
(.XXXXX).

Y PL - SAME AS
RECEIPT OF
PROCEED OR NEW
X PL FOR Y
COMPONENT.

Z PL - SAME AS
Y PL FOR Z
COMPONENT.

MONITOR DSKY:
OBSERVE VERR-NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
OF PLANET POSITION
VECTOR.

ARE THE POSITION
VECTOR COMPONENTS
CORRECT?

Y

#450

#460

#470

++
+00
++
PCN
669

RESET SIGHTING MARK
FLAG.

•
•
•
•
•
•
•
•
•
•

• • •

X I T
Q 5 3

1-5

CHANGE CONTROL NOTES

REV 06 PCR MIT 66
PCR MIT 67
REV 07 PCR 206
PCR 448
REV 08 PCR 206 EDITORIAL
REV 09 PCN 668, EDITORIAL

00597000
00598000
00599000
00600000
00600000
00600001
00600002

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SIGHTING DATA DISPLAY ROUTINE (R54)

LOGIC REV 07 11/27/68

PURPOSE: (1) TO TEST THE ACCURACY OF A PAIR OF CELESTIAL BODY SIGHTINGS.

ASSUMPTIONS: (1) THE ROUTINE IS NORMALLY AUTOMATICALLY SELECTED BY THE IMU ORIENTATION DETERMINATION PROGRAM (P51), BY THE IMU REALIGN PROGRAM (P52), BY THE BACK-UP IMU ORIENTATION DETERMINATION PROGRAM (P53), OR BY THE BACK-UP IMU REALIGN PROGRAM (P54).

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
--------------	-----	--------	------	-----------	------	---------------

.CMC
.ROUTINE
.SELECTION

...
.

START SIGHTING DATA
DISPLAY ROUTINE
(R54).

.
.
.

CALCULATE ANGLE BE-
TWEEN TWO CELESTIAL
BODIES USING STORED
EPHEMERIS DATA
(ACTUAL)

.
.
.

CALCULATE ANGLE BE-
TWEEN TWO CELESTIAL
BODIES USING CELES-
TIAL BODY VECTORS
DERIVED FROM MARK
ANGLES (INDICATED)

.
.
.
.
.
.
.
.
.
.
.

#10

#20

#30

R54/COLOSSUS
R54/SUNDANCE
R54/SUNDANCE

R54/COLOSSUS
R54/SUNDANCE
R54/LUMINARY

TERMINATE FLASH UPON
RECEIPT OF PROCEED
OR RECYCLE

.....

KEY IN RECYCLE
V32E

.P

.R

.D

.C

.E

.F

.D

.

.

.

...

EXIT "A"
FROM R54
CONTINUE AS
DEFINED IN
CALLING PRO-
GRAM/ROUTINE

.R

.E

.C

.Y

.C

.L

.F

.

.

.

...

EXIT "B"
FROM R54
CONTINUE AS
DEFINED IN THE
CALLING PRO-
GRAM/ROUTINE

.

.

...

.

EXIT "B"
FROM R54
CONTINUE AS
DEFINED IN THE
CALLING PRO-
GRAM/ROUTINE

#90

#100

#110

CHANGE CONTROL NOTES

REV 07 EDITORIAL

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PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
--------------	-----	--------	------	-----------	------	---------------

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH AND DISPLAY OF
GYRO TORQUING ANGLES

#20

R55/COLOSSUS
R55/SUNDANCE
R55/LUMINARY

GYRO TORQUING
ANGLES-THE ANGLE
THRU WHICH EACH
GYRO MUST BE
TORQUED TO COM-
PLETE THE FINE
ALIGNMENT. ALL
ANGLES IN DEGREES
TO NEAREST .001
DEGREE.

#30

#40

SHALL I PERMIT
TORQUING? CONSIDER
MAGNITUDE OF TORQU-
ING ANGLES.

#50

N. Y.
.
.
.

WAIT FOR KEYBOARD
ENTRY

KEY IN
RECYCLE
V32F

#60

EXIT R55

#70

TERMINATE FLASH
UPON RECEIPT OF
PROCEED OR RECYCLE

KEY IN PROCEED

EXIT R55

#80

```

      .
      .
      .
      .
      .
      .
      .
EXIT R55
      .
      .
      .
      .
-----
PULSE IRIGS THROUGH
DESIRED ANGLES
-----
      .
      .
      .
      .
      .
      .
      .
      .
      .
      .
EXIT R55

```

#100

LOGIC REV 03 PCR MIT 66

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10010 354 10 12/26/62

PROG CONT	CMD	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
--------------	-----	--------	------	-----------	------	---------------

ISSUE 4/2004

#50

#70

●●●

DEB/FOLOSIUS

#90

WAIT FOR KEYBOARD
ENTRY

KFY IN ENTER

TERMINATE FLASH UPON
RECEIPT OF ENTER

#100

- E
- N
- T
- F
- O

STORE THREE FOUR
ANGLES AND THE TWO
OPTICS ANGLES OF
NOUN 94

#110

HOI D
.....
CNA P

```
FLASH VERB NOUN
TO REQUEST PLEASE
PERFORM TERMINATION
OF THIS ROUTINE
V50 N25
R1 00016
R2 BLANK
R3 BLANK
```

MONITOR DSKY:
OBSERVE VERR NOUN
FLASH TO REQUEST
PLEASE PERFORM TERM-
INATION OF THIS
ROUTINE.

#120

WAS THE SIGHTING
SATISFACTORY?

#130

• Y	• H
•	•
•	•

WAIT FOR KEYBOARD
ENTRY

KEY IN ENTERED

4143

TERMINATE FLASH HODN
 RECEIPT OF ENTER OR
 PROCEED

KEY IN PROCEED

.F .D
 .N .D
 .T .D
 .E .C
 .D .E
 . .F
 . .D

ERASE MARK
 DATA

#150

#160

HOLD .

 SNAP .

FLASH VERB-NOUN TO
 REQUEST RESPONSE AND
 DISPLAY OF CELESTIAL
 BODY CODE

V01 N71
 01-CELESTIAL BODY
 CODE
 02-PLANET
 03-PLANET

01-CELESTIAL BODY
 CODE-THE DESIGNATION
 OF THE CELESTIAL
 BODY WHICH WAS
 MARKED (IN TOTAL).

00 - PLANET (ANY
 PLANET)
 01/45 - STAR (FROM
 CELESTIAL BODY
 CODE LIST)
 46 - SUN
 47 - EARTH
 50 - MOON

MONITOR DSKY:
 OBSERVE FLASHING
 VERB-NOUN TO REQUEST
 RESPONSE AND DISPLAY
 OF CELESTIAL BODY
 CODE.

AM I SATISFIED
 WITH THIS CODE?

.Y .N

#170

#180

#190

 WAIT FOR KEYBOARD
 ENTRY

 KEY IN
 PROCEED

#200

 TERMINATE FLASH UPON
 RECEIPT OF PROCEED
 OR NEW DATA.

 KEY IN V21E AND
 LOAD STAR CODE:

 CELESTIAL BODY
 CODE - CELESTIAL
 BODY DESIGNATION
 FROM CARRY ON
 DATA.

#210

 .P .NEW
 .R .DATA
 .D
 .C
 .E
 .F
 .D

 STORE
 NEW
 DATA

#220

 IS CELESTIAL BODY
 CODE 00?

 .N .Y

 IS CELESTIAL
 BODY CODE 46,
 47 OR 50?

 .N .Y

 IS THE TARGET A STAR
 OR THE EARTH MOON OR
 SUN?

 .N .Y

 EXIT

#230

#240

#250

```

OBTAIN STAR
VECTOR FROM
STORED
EPHEMERIS

```

#260

```

CALCULATE CEL-
ESTIAL BODY
VECTOR FOR THE
BODY DEFINED
BY THE STAR
CODE.

```

#270

```

POSS
HOLD
.....
SNAP

```

```

FLASH VERR-
NOUN TO RE-
QUEST RESPONSE
AND DISPLAY
PLANET POSI-
TION VECTOR:
V06ARR
R1-X PL
R2-Y PL
R3-Z PL

```

```

MONITOR DSKY:
ORSEPRV VERR-NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
OF PLANET POSITION
VECTOR.

```

#280

```

++
+10
++
EDIT

```

```

Y PL - THE Y
COMPONENT OF 1/2
UNIT POSITION
VECTOR OF THE
PLANET AT GET
IN REFERENCE
COORDINATES.
TO THE FIFTH
PLACE
(.XXXXXX).

```

```

ARE THE POSITION
VECTOR COMPONENTS
CORRECT?

```

#290

```

Y PL - SAME AS
X PL FOR Y
COMPONENT.

```

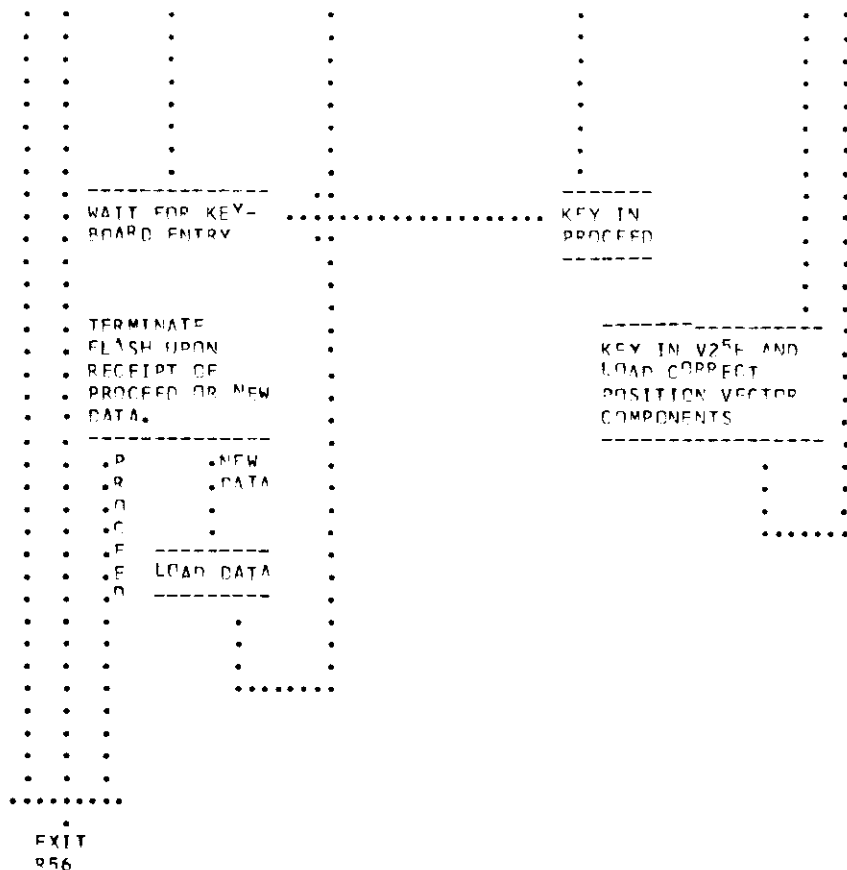
```

Z PL - SAME AS
X PL FOR Z
COMPONENT.

```

#300

133



856/COLLOSSUS

#310

#320

#330

CHANGE CONTROL NOTES

REV 07 RCR MIT 66
 REV 03 RCR 487
 REV 02 RCR 206 EDITORIAL
 REV 01 EDITORIAL

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OPTICS CALIBRATION ROUTINE (R57)

LOGIC REV 09 04/12/68

PURPOSE: (1) TO MEASURE THE EFFECT OF SOLAR RADIATION ON THE SXT TRUNNION ANGLE.
 (2) TO STORE THE MEASURED TRUNNION BIAS FOR USE BY THE CISLUNAR NAVIGATION PROGRAM (P23).

ASSUMPTIONS: (1) TRUNNION BIAS IS MEASURABLE ONLY IN SPACE (VACUUM) .
 (2) TRUNNION BIAS IS A FUNCTION OF THE ANGLE BETWEEN THE LANDMARK LINE OF SIGHT AND THE SUN AND IS CONSTANT OUTSIDE THE CONE FROM PLUS OR MINUS 15 DEGREES FROM THE LANDMARK LINE OF SIGHT.
 (3) AFTER THE FIRST TIME THIS ROUTINE IS PERFORMED DURING A MISSION IT SHOULD NOT BE NECESSARY TO REPEAT IT IF THE SUN ANGLE WAS OUTSIDE THE CONE OF ASSUMPTION 2 AND WILL BE OUTSIDE THE CONE DURING THE NAVIGATION MARKS TO BE TAKEN DURING P23. IF THE SUN ANGLE IS WITHIN THIS CONE THIS ROUTINE SHOULD BE REPEATED EACH TIME THROUGH P23.
 (4) THE OPTICS SHOULD "SOAK" IN AN ATTITUDE WITHIN THE REGION THAT WILL BE USED FOR P23 FOR AT LEAST A HALF HOUR PRIOR TO CALIBRATION AND MARKING ON THE NAVIGATION TARGETS OF P23.

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
--------------	-----	--------	------	-----------	------	---------------

.CMC
 .ROUTINE
 .SELECTION

.

...

.

 START OPTICS CALI-
 BRATION ROUTINE
 (R57)

#10

.....

.....

.....

.....

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.....

.....

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.....

.....

.....

.....

.....

.....

.....

.....

.....

.....

.....

.....

.....

.....

.....

.....

.....

.....

HOLD . FLASH VERB TO RE-
 QUEST PLEASE PERFORM
 SNAP . CALIBRATION MARK
 V59N-BLANK
 R1-BLANK
 R2-BLANK
 R3-BLANK

MONITOR DSKY:
 OBSERVE VERB 59
 FLASH TO REQUEST
 PLEASE PERFORM
 CALIBRATION MARK

#20

WAIT FOR KEYBOARD
 ENTRY

HAVE I PREVIOUSLY
 CALIBRATED THE
 OPTICS?

#30

.Y .N

WILL THE OPTICS
 BE USED IN THE
 SAME REGION THAT
 I CALIBRATED
 THEM FOR PRE-
 VIOUSLY? (REFER

#40

TO ASSUMPTIONS
 3 AND 4)

#50

.Y .N

KEY IN ENTER

TERMINATE FLASH UPON
 RECEIPT OF MARK OR
 ENTER

PERFORM OPTICS CALI-
 BRATION PER CHECK-
 LIST AND MARK

#60

.MARK .ENTER

#70

#80

#90

#100

#110

#120

KFY IN
PROCEED

#140

#150

#160

652

ATTITUDE MANEUVER ROUTINE (R6J)

LOGIC REV 11 11/27/68

- PURPOSE: (1) TO MANEUVER THE LM/CSM OR CSM ALONE TO AN ATTITUDE SPECIFIED BY THE PROGRAM IN PROGRESS.
- ASSUMPTIONS: (1) THE FINAL ATTITUDE DESIRED, DEFINED AS FOLLOWS, HAS BEEN STORED BY THE CALLING PROGRAM:
- (A) A SPECIFIC BODY FIXED VECTOR AND A DIRECTION IN SPACE TO WHICH THIS VECTOR IS TO BE ALIGNED (THE 3-AXIS FLAG IS RESET).
 - (R) A THREE AXIS (ORTHOGONAL) INERTIAL ORIENTATION TO WHICH THE THREE BODY AXIS ARE TO BE ALIGNED (THE THREE AXIS FLAG IS SET).
- (2) THE MANEUVER MAY BE PERFORMED AUTOMATICALLY BY THE GNCS OR PERFORMED MANUALLY WITH AN OPTIONAL FINAL AUTOMATIC GNCS CONTROLLED TRIM MANEUVER. THIS OPTIONAL TRIM MANEUVER SHOULD BE CONSIDERED ESSENTIAL FOR MANEUVERS TO SPS THRUSTING ATTITUDES.
- (3) THE DAP DATA LOAD ROUTINE (R03) HAS BEEN PERFORMED PRIOR TO THIS ROUTINE.
- (4) THE ROUTINE IS AUTOMATICALLY SELECTED BY THE PROGRAM OR ROUTINE REQUIRING THE ATTITUDE MANEUVER.
- (5) IF THIS ROUTINE WAS SELECTED BY THE TRACKING ATTITUDE ROUTINE (R61) THE V50N18 AND THE V06N18 IN THIS ROUTINE ARE PRIORITY DISPLAYS. THE V50N18 DISPLAY WILL REMAIN UP A MINIMUM OF 2 SECONDS. RESPONSE AFTER 2 SECONDS WILL CAUSE THE PROGRAM TO CONTINUE AS DESCRIBED

PRG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
	.CMC .ROUTINE .SELECTION					
	----- START ATTITUDE MANEUVER ROUTINE (R60) -----					#10
	. . .					
	----- OBTAIN DESIRED ATTITUDE SPECIFICA- TION FROM THE CALL- ING PROGRAM -----					#20
	. . .					
						R60/COLOSSUS R60/LUMINARY

• N • Y

• •

• •

IS THE 3 AXIS FLAG
SET?

CALCULATE FINAL
VEHICLE ATTITUDE
TO MEET THE
DESIRED ATTITUDE
SPECIFICATION
(VECPPOINT ROU-
TINE). THIS FINAL
VEHICLE ATTITUDE
WILL BE CALCU-
LATED TO MEET THE
ATTITUDE SPECIFI-
CATION IN SUCH A
WAY AS TO CON-
SERVE RCS FUEL
AND NOT CONSTRAIN
ANY UNSPECI-
FIED DEGREE OF
FREEDOM.

#60

.....

#70

- SELECT GIMBAL
- ANGLES CORRESPONDING TO
- PREFERRED
- VEHICLE ATTITUDE AND PRESENT IMU ORIENTATION

#80

```

HOLD. . . . . FLASH VERB-NOUN TO
..... REQUEST PLEASE
SNAP. . . . . PERFORM AUTO
          MANEUVER:
          V50 N18
          R1-CG ROLL
          R2-IG PITCH
          R3-MG YAW

```

OG - FINAL DESIRED
OUTER GIMBAL ANGLE
IN DEGREES TO
NEAREST .01 DEGREES.

IG - FINAL DESIRED
INNER GIMBAL ANGLE
IN DEGREES TO
NEAREST .01 DEGREES

MG - FINAL DESIRED
MIDDLE GIMPAL ANGLE
IN DEGREES TO
NEAREST .01 DEGREES.

WAIT FOR KEYBOARD
ENTRY

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST
PLEASE PERFORM AUTO
MANEUVER AND DISPLAY
OF DESIRED GIMBAL
ANGLES.

#90

REVIEW THE PRESENTLY
DISPLAYED GIMBAL
ANGLES AND THE
PRESENT ATTITUDE. AM
I WITHIN THE PRESENT
RCS DAP DEADBAND
LIMITS IN EACH AXIS?

#100

#110

.N .Y

.

.

DO I WISH
TO FURTHER
ADJUST THE
VEHICLE
ATTITUDE
ABOUT THE

#120

#130

DESIRED
VECTOR?
(NOT POS-
SIBLE FOR
ALL CASES.
SEE ASSUMP-
TION 1)

.N .Y

#140

TERMINATE FLASH UPON
RECEIPT OF ENTER OR
PROCEED

KEY IN ENTER

.P .F
.R .N
.Q .T
.C .F
.E .R
.E
.D

EXIT

#150

RESET 3-
AXIS FLAG.

EXIT

#160

SHALL I HAVE THE
GNCS PERFORM THE
MANEUVER AUTO-
MATICALLY?

.Y .N

#170

IS THE 3 AXIS FLAG
SET?

.N .Y

SELECT CMC
CONTROL AND
SELECT THE
AUTO MODE.

 CALCULATE FINAL
 VEHICLE ATTITUDE
 TO MEET THE
 DESIRED ATTITUDE
 SPECIFICATION
 (VECPPOINT ROU-
 TIME). THIS FINAL
 VEHICLE ATTITUDE
 WILL BE CALCU-
 LATED TO MEET THE
 ATTITUDE SPECIFI-
 CATION IN SUCH A
 WAY AS TO CON-
 SERVE RCS FUEL
 AND NOT CONSTR-
 RAIN ANY UNSPEC-
 IFIED DEGREE OF
 FREEDOM.
 NOTE: GNCS CAP-
 ABILITY TO PER-
 FORM MANEUVER
 AUTOMATICALLY
 WILL BE COMPROM-
 ISED IF THE
 ATTITUDE IS
 CHANGED BY MANUAL
 INPUTS AFTER THIS
 TIME

 SELECT GIMBAL
 ANGLES CORRES-
 PONDING TO
 PREFERRED
 VEHICLE ATTIT-
 UDE AND PRES-
 ENT IMU ORIENT-
 TATION

 SHALL I HAVE
 THE GNCS
 RECOMPUTE THE
 DESIRED ATTIT-
 UDE WITHOUT
 PERFORMING
 THE AUTOMATIC
 MANEUVER?
 (NOT POSSIBLE
 FOR ALL
 CASES. SEE
 ASSUMPTION 1)

.Y .N

 EITHER SELECT
 SCS CONTROL
 OR PLACE
 MODE SWITCH
 NOT IN AUTO.

..... KEY IN
 PROCEED

#180

#190

#200

#210

#220

PERFORM
ATTITUDE
MANEUVER
MANUALLY
USING
RHC AND
BY REFF-
ERENCE TO
THE OUT
THE WIN-
DOW VIEW
AND/OR
THE FDAI
BALL AND
ATTITUDE
ERROR
NEEDLES.

#230

#240

#250

IS S/C CONTROL CMC?

.N .Y

DID I DIRECT THE CMC
TO PERFORM THE MANE-
UVER AUTOMATICALLY?

.Y .N

#260

IS THE AUTO
MODE SELEC-
TED?

.N .Y

#270

130

TRACKING ATTITUDE ROUTINE (R61)

LOGIC REV 16 12/26/68

- PURPOSE: (1) TO COMPUTE THE PREFERRED TRACKING ATTITUDE OF THE CSM WHICH ENABLES OPTICS TRACKING OF THE LM AND LM TRACKING OF THE CSM RADAR TRANSPONDER AND TO COMPUTE THE +X-AXIS TRACKING ATTITUDE OF THE CSM WHICH ENABLES COAS TRACKING OF THE LM.
- PCR
606
++
+16
++
- ASSUMPTIONS: (1) THE PREFERRED TRACKING ATTITUDE IS DEFINED AS FOLLOWS:
- (A) THE TRACK AXIS (I) IS ALIGNED ALONG THE LOS TO THE LM. THE TRACK AXIS (I) IS DEFINED AS:
- $$\text{UNIT (I)} = \text{UNIT (Z)} \cos 55 \text{ DEG} + \text{UNIT (X)} \sin 55 \text{ DEG}$$
- (B) THE CSM ORIENTATION ABOUT THE TRACK AXIS (I) IS A FUNCTION OF THE EXISTING ATTITUDE AT THE TIME OF THE CALCULATION AND IS CALCULATED SO AS TO YIELD A MINIMUM ATTITUDE MANUEVER.
- (2) THE ROUTINE IS AUTOMATICALLY CALLED BY THE RENDEZVOUS NAVIGATION PROGRAM (P20).
- (3) THE GIMBAL ANGLES REQUIRED TO POINT THE PREFERRED LOS AT THE LM ARE AVAILABLE BY KEYING IN V16N95F.
- (4) THE GIMBAL ANGLES REQUIRED TO POINT THE +X-AXIS LOS AT THE LM ARE AVAILABLE BY KEYING IN V16N96F.
- (5) IF THE SURFACE FLAG HAS BEEN SET (V44F) THIS ROUTINE WILL ASSUME THE LM TO BE AT THE MOST RECENTLY DEFINED LANDING SITE. THE LANDING SITE CAN BE PLACED INTO THE CMC DURING THE ORBIT NAVIGATION PROGRAM (P22), OR BY UPLINK

PRG
CONT

CMC

GROUND

CREW

CHECKLIST

TIME

TOTAL
TIME

.CMC
.ROUTINE
.SELECTION
.
.
.

START PREFERRED
TRACKING ATTITUDE
ROUTINE (R61)

.
.
.
.

#10

1A

• Y	• N
•	•
•	•

[illegible]

```

.
.
.
.
.
.
.
.
.
EXIT

```

• N

[illegible]

#60

137

•
•
•
•
•
•

EXTRAPOLATE LM
AND CSM STATE
VECTORS TO
PRESENT TIME
USING CONIC
EQUATIONS

•
•
•

CALCULATE THE PRE-
FERRED TRACKING
ATTITUDE FROM CSM
TO LM. (PREFERRED
UNIT VECTOR ALIGN-
ED WITH LCS FROM
CSM TO LM). THIS
ATTITUDE WILL BE
COMPUTED (VECPNT) TO
POINT THE PRE-
FERRED AXIS AT THE
LM BUT WILL NOT
CONSTRAIN THE NON-
CRITICAL ORIENTA-
TION ABOUT THAT
VECTOR.

•
•
•

COMPUTE REQUIRED
GIMBAL ANGLES, AT
THE PREFERRED
TRACKING ATTITUDE
IF THE PRESENT IMU
ORIENTATION IS
USED AND STORED
IN NCUM 05

•
•
•

CALCULATE +X-AXIS
TRACKING ATTITUDE
FROM CSM TO LM (+X-
AXIS ALIGNED WITH
LCS FROM CSM TO
LM). THIS ATTITUDE
WILL BE COMPUTED
(VECPNT) TO
POINT THE +X-AXIS

PA1/COLOSSUS

#70

#80

#90

#100

#110

#120

1A

PA1/COLOSSUS

•

#130

•

•

•

#140

• Y
•

#150

•

•

•

#160

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• www.pearsoned.com

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•	•
•	•

4270

DO ATTITUDE MANIPULATED
ROUTINE (R60).

1A

(3) AMOUNTS BY WHICH THE COUNTER-DESIGNED REGISTERS SHOULD BE INCREMENTED AT 0.1 SECOND INTERVALS.

#347

```

      .  

      .  

      .  

      ...  

      .  

EXIT RSL
```

● 350

10010	RFV 11	PCR 4	MIT 54
	RFV 12	PCR	MIT 66
	RFV 13	PCR 206	
		PCR 225	
	RFV 14	PCR 507	
	RFV 15	PCR 521	
	RFV 16	PCR 606	

PURPOSE: (1) TO PROVIDE THE CREW WITH THE ABILITY TO SPECIFY A FINAL VEHICLE ATTITUDE FOR USE BY A CMC-CONTROLLED ATTITUDE MANEUVER.

ASSUMPTIONS: (1) THE ROUTINE IS MANUALLY SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

(2) THE DAP DEADBAND DURING THIS ROUTINE IS AS DEFINED BY THE LOAD DAP DATA ROUTINE (R03).

(3) THIS ROUTINE CAN ONLY BE ENTERED FROM THE CMC IDLING PROGRAM (POO).

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
			. CREW . PROGRAM . SELECTION			
	----- START CREW-DEFINED MANEUVER ROUTINE (R62) ----- . : : : :	.	----- KEY IN V49E ----- . : : : : :			#10
	----- IS CURRENT PROGRAM POO? ----- .Y .N . .	.	----- MONITOR DSKY: DOES OPERATOR ERROR LIGHT COME ON INDICATING THAT THIS ROUTINE CANNOT BE SELECTED AT THIS TIME? ----- .Y .N .			#20
++ + +05 + ++ PCN 586	----- IS ANOTHER EX- TENDED VERB ACTIVE? ----- .N .Y. .	.				

R62/COLOSSUS

#30

TURN ON OPER-
ATOR ERROR
LIGHT

KEY V37E00E
AND THEN KEY
V49E

EXIT

EXIT

#40

HOLD .. FLASH VERB-NOUN TO
..... REQUEST RESPONSE AND
SNAP .. DISPLAY FINAL GIMBAL

ANGLES:

V06 N22
R1- CG ROLL
R2- IG PITCH
R3- MG YAW

ALL GIMBAL ANGLES IN
DEGREES TO NEAREST
.01 DEGREE.

WAIT FOR KEYBOARD
ENTRY

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST RE-
SPONSE AND DISPLAY
OF FINAL GIMBAL
ANGLES.

DO I WISH TO KEY IN
NEW GIMBAL ANGLES TO
BE USED BY ROUTINE
R60?

.N .Y

KEY IN V25E AND
LOAD NEW GIMBAL
ANGLES

#50

#60

#70

TERMINATE FLASH UPON
RECEIPT OF PROCEED
OR NEW DATA

KEY IN PROCEED

.NEW .P
.DATA .R
. .C

STORE .E
NEW .F

#80

621

R62/COLOSSUS

#90

#100

#110

DATA

SET 3 AXIS FLAG FOR
USE BY ATTITUDE
MANEUVER ROUTINE
(R60)

DO ATTITUDE MANEUVER
ROUTINE (R60)

EXIT R62

DO ATTITUDE MANEUVER
ROUTINE (R60)

EXIT R62

CHANGE CONTROL NOTES

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RENDEZVOUS FINAL ATTITUDE ROUTINE P63

LOGIC REV 13 11/27/68

PURPOSE:

- (1) TO CALCULATE THE FINAL GIMBAL ANGLES REQUIRED TO POINT THE CSM +X AXIS AT THE LM AND TO CALCULATE THE FINAL GIMBAL ANGLES REQUIRED TO POINT THE PREFERRED TRACKING AXIS AT THE LM.
- (2) TO DISPLAY THE GIMBAL ANGLES CORRESPONDING TO THE ATTITUDE OPTION SELECTED BY THE ASTRONAUT.
- (3) TO CALL THE ATTITUDE MANEUVER ROUTINE (P60) FOR AUTOMATIC MANEUVER CAPABILITY.

ASSUMPTIONS:

- (1) THE PREFERRED TRACKING ATTITUDE IS DEFINED AS FOLLOWS:
(A) THE TRACK AXIS (1) IS ALIGNED ALONG THE LOS TO THE LM. THE TRACK AXIS (1) IS DEFINED AS:

$$\begin{matrix} \text{UNIT}(1) = \text{UNIT}(2) \cos 55 \text{ DEG} + \text{UNIT}(X) \sin 55 \text{ DEG} \\ \text{---} \quad \text{---} \quad \text{---} \\ \text{---} \quad \text{---} \quad \text{---} \\ \text{---} \quad \text{---} \quad \text{---} \\ \text{---} \quad \text{---} \quad \text{---} \end{matrix}$$

- (B) THE CSM ORIENTATION ABOUT THE TRACK AXIS (1) IS A FUNCTION OF THE EXISTING ATTITUDE AT THE TIME OF THE CALCULATION SO AS TO YIELD A MINIMUM ATTITUDE MANEUVER.

- (2) TO SAVE TIME THE CSM ATTITUDE CONTROL MODE SHOULD BE PRESELECTED (FOR AUTOMATIC MANEUVERS P03 SHOULD HAVE BEEN DONE AND THE CMC AUTO MODE SELECTED).
- (3) THIS ROUTINE MAY BE SELECTED IN P00 ONLY.
- (4) THIS ROUTINE IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

PRG
CONT

CMC

GROUND

CREW

CHECKLIST

TIME

TOTAL
TIME

START RENDEZVOUS
FINAL ATTITUDE
ROUTINE P63

KEY IN VR0F

IS THE CURRENT PRO-
GRAM P002

YES. NO

• •
• •
• •
• •

• CREW
• ROUTINE
• SELECTION
•
•
•

#10

#20

P63/C01/DSSUS

++
+
+
+13
+
++
PCN
586

IS ANOTHER EX-
TENDEO VERR
ACTIVE?

N. Y.

TURN ON
OPERATOR ERROR
LIGHT

EXIT
R63

MONITOR DSKY:
DOES OPERATOR
ERROR LIGHT COME ON,
INDICATING THAT THIS
ROUTINE CAN NOT BE
SELECTED AT THIS
TIME?

.Y .N

IN ORDER TO TURN
THIS ROUTINE ON
SELECT CMC IDLING
PROGRAM (P00) BY
KEYING V37E00F
AND RESELECT THIS
ROUTINE

EXIT
R63

DO IMU STATUS CHECK
ROUTINE (R02)

DO IMU STATUS CHECK
ROUTINE (R02)

#30

#40

#50

#60

#70

#130

RESET THE	SET THE
PREFERRED	PREFERRED
ATTITUDE	ATTITUDE
FLAG.	FLAG.

#140

•	•
•	•
•	•
•	•
•	•
•	•

```
PCN      . CSM STATE VECTORS
531      . FORWARD TO THE
++       . PRESENT TIME +1 MIN
+12      . USING CONIC EQUATIONS
++       . -----
```

#150

CALCULATE THE PREFERRED TRACKING ATTITUDE FROM CSM TO LM. (PREFERRED UNIT VECTOR ALIGNED WITH LOS FROM CSM TO LM). THIS ATTITUDE WILL BE COMPUTED (VECPNT) TO POINT THE PREFERRED AXIS AT THE LM BUT WILL NOT CONSTRAIN THE NON-CRITICAL ORIENTATION ABOUT THAT VECTOR.

***160**

#170

- COMPUTE REQUIRED
- GIMBAL ANGLES AT THE
- PREFERRED TRACKING
- ATTITUDE IF THE
- PRESENT IMU ORIENT-
- TATION IS HELD AND
- SLOPE IN NOON 95.

•	•
•	•
•	•

#190

- CALCULATE THE +Y-
- AXIS TRACKING ATT-
- ITUDE FROM CSM TO LM
- (+X-AXIS ALIGNED
- WITH LOS FROM CSM TO
- LM). THIS ATTITUDE
- WILL BE COMPUTED
- (VECPNT) TO POINT
- THE +X-AXIS AT THE
- LM BUT WILL NOT CON-
- STRAIN THE NON-CRIT-
- ICAL ORIENTATION
- ABOUT THAT VECTOR
- (ROLL).

#200

•	•
•	•
•	•

- COMPUTE REQUIRED
- GIMBAL ANGLES AT
- THE +X-AXIS TRACKING
- ATTITUDE IF THE
- PRESENT IMU ORIENT-
- TATION IS HELD AND
- STORE IN NOUN 96.

#210

■	■
●	●
●	●

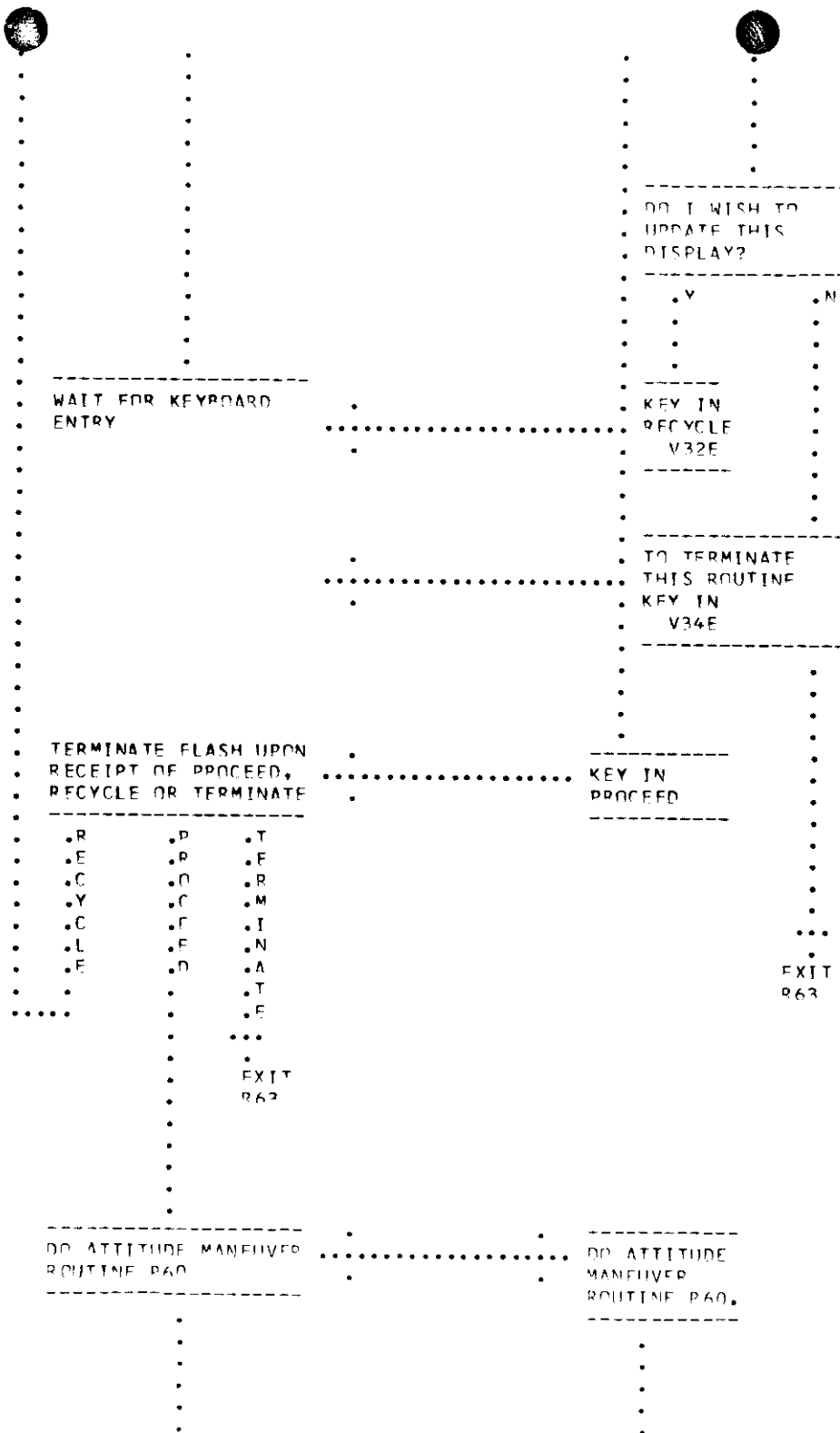
#220

- IS THE PREFERRED
- ATTITUDE FLAG SET
- (SEE P2012)

[illegible]

[illegible]

#2RQ



#290

#300

#310

#320

#330

REV 10 PCR MIT 66
REV 11 PCR MIT 206
REV 12 PCN 531
REV 13 PCR 206 EDITORIAL
PCN 586

FRESH START VERB (V36)

LOGIC REV 04 04/10/68

PURPOSE: (1) TO INITIATE A COMPUTER FRESH START

ASSUMPTIONS: (1) FRESH START IS CREW INITIATED BY DSKY ENTRY.

(2) ALTHOUGH THE REFSMMAT FLAG IS RESET BY A FRESH START INDICATING THAT THE REFSMMAT IS NOT GOOD, THE ACTUAL REFSMMAT IS NOT CHANGED.

(3) FRESH START RESETS THE CSM MOON FLAG AND THE LM MOON FLAG INDICATING EARTH ORBIT FOR STATE VECTOR INTEGRATION.

(4) IF A FRESH START INTERRUPTS STATE VECTOR INTEGRATION, THE STATE VECTOR MAY BE INVALIDATED.

(5) THIS PROCESS MAY BE SELECTED AT ANY TIME.

++
+04
+
+04
++PROC
CONT

CMC

GROUND

CREW

CHECKLIST

TIME

TOTAL
TIMECREW
SELECTION.
.
.-----
START CREW INITIATED
FRESH START
-----.

KEY IN V36E

#10

.
.
.-----
ZERO OUTBIT CHANNELS
5 (RCS PITCH AND
YAW) AND 6 (RCS
ROLL)
-----++
+04
+04
++.
.
.

#20

SET TIME 3 = 37777
TIME 4 = 37775
TIME 5 = 37774
-----.
.
.
.
.
.

#30

V36/COLOSSUS

.
 .
 .
 .

 ZERO OUTBIT CHAN-
 NELS: 11 (EXCEPT
 ENGINE ON, OFF AND
 ISS WARNING);
 12 (EXCEPT COARSE
 ALIGN ENABLE, ZERO
 IMU CDU'S, ENABLE
 IMU ERROR COUNTER;
 13 (EXCEPT TELEMETRY
 BITS, RESET TRAP
 BITS AND TARUPT
 BIT);
 14 (EXCEPT CYRO EN-
 ABLE).

++
 +04
 ++

#40

.
 .
 .

 TERMINATE WAITLISTED
 TASKS

#50

.
 .
 .

 CLEAR ALL EXECUTIVE
 REGISTER SFTS

#60

.
 .
 .

 INDICATE NO ACTIVE
 JOBS

.
 .
 .

 MAKE ALL VAC AREAS
 AVAILABLE

#70

.
 .
 .

 BLANK DSKY REGISTERS
 (PROGRAM, VERR,
 NCUN, R1, P2, R3.)

#80

.
 .
 .
 .
 .
 .

 RESET DISPLAY/
 ASTRONAUT INTERFACE
 FLAGS

#90

.
 .
 .

 TURN OFF DSKY
 DISCRETE LIGHTS.

.
 .
 .

 CLEAR FAIL REGIS-
 TERS, SELF CHECK
 ERROR COUNTER, AND
 RESTART COUNTER

#100

.
 .
 .

 CLEAR SELF-CHECK
 ERROR REGISTERS,
 MODE REGISTER.

#110

++
 +04
 ++

.
 .
 .

 INITIALIZE PIPA AND
 TELEMETRY FAIL FLAGS

#120

.
 .
 .

 ZERO OUTBIT CHAN-
 NELS: 11("AM"
 RELAYS); 12 (GNC);
 13 (AGC); AND
 14 (ISS)

#130

.
 .
 .

 INITIALIZE DOWNLINK
 WITH P00 DOWNLIST

#140

•
•
•

#150

•

#160

•

#170

•

#180

•

•

•

EXIT

435

CHANGE CONTROL NOTES

LOGIC REV 04 PCR MIT 66

00231000

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LOGIC DEY 03 07/09/68

ASSUMPTIONS: (1) THE PROCESS IS CREW SELECTED BY DSKY ENTRY.

(2) THE PROCESS MAY NOT BE SELECTED IF THE ISS IS IN THE COARSE ALIGN MODE AND IN GIMBAL LOCK.

(3) THE PROCESS IS INTENDED PRIMARILY FOR USE ON THE GROUND.

V4 ON20 / COLD SUS

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 •
 •
 •
 •
 •
 •
 •

•
 •
 •
 ••••
 •
 ΓΥΤΤ

THRM OFF NO ATT
LIGHT

READ ISS COUN
READ COUNTRIES

ENTER? TIME
ALIGN MODE

+ +
 + 0 2
 + 7 2
 + +

639

.
.
.
.
.
EXIT

V4.0N20/COLLOSSUS

CHANGE CONTROL NOTES

REV 03 PCB 206

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PURPOSE: (1) TO ALIGN THE IMU TO GIMBAL ANGLES SPECIFIED BY THE ASTRONAUT.

(2) TO COARSE ALIGN TO 0,0,0 WHEN IN GIMBAL LOCK AND COARSE ALIGN.

ASSUMPTIONS:

- (1) THE PROCESS IS CREW SELECTED BY DSKY ENTRY.
- (2) THE PROCESS MAY BE SELECTED ONLY WHEN NO OTHER EXTENDED VERB IS ACTIVE.
- (3) THE ACCURACY OF THE ALIGNMENT IS TESTED TO A TOLERANCE OF +2 DEGREES. IF THIS TOLERANCE IS EXCEEDED, THE ASTRONAUT WILL BE NOTIFIED VIA A PROGRAM ALARM.
- (4) V41 MAY ONLY BE USED WITH N20 OR N91. REFER TO OPTICS COARSE ALIGN EXTENDED VERB (V41N91).
- (5) THE PROCESS SHOULD NEVER BE USED DURING A PROGRAM WHICH REQUIRES THAT THE IMU BE ON AND ALIGNED.

[illegible]

IS THE IMU
BEING
INITIALIZED?

.N .Y

TURN ON OPR
ERR LIGHT

EXIT

++
+04
+
+04
++
EDIT

HOLD .
.....
.

FLASH VERB-NOUN TO
REQUEST LOAD OF
DESIRED CDU ANGLES.

V2IN22
R1-OG ROLL
R2-IG PITCH
R3-MG YAW

ALL REGISTERS
INITIALLY BLANK

ALL ANGLES IN DE-
GREES TO NEAREST .01
DEGREE

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH REQUESTING
LOAD OF CDU ANGLES

SHALL I LOAD CDU
ANGLES?

.Y .N

#30

#40

#50

#60

#70

READ PRESENT IMU OR-
IENTATION W.R.T. THE
VEHICLE.
(GIMBAL ANGLES)

ARE THE GIMBALS
WITHIN 2 DEGREES OF
THE DESIRED ANGLES?

.Y .N

- TURN ON PROGRAM
- ALARM AND STORE
- ALARM CODE
- (00211)

EXIT

MONITOR DSKY:
DOES PROGRAM ALARM
INDICATE THAT THE
IMU GIMBALS DID NOT
DRIVE TO WITHIN 2
DEGREES OF THE DE-
SIRED ANGLES?

```

.N          .Y
.           .
.           .
...        .
.           .
EXIT       .

```

CHECK STATUS OF CW
PANEL. IS AN ISS
MALFUNCTION
INDICATED?

• Y • N

#160

#170

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++
+05
++
EDIT

#40

IS OPTICS MODE
CMC?

.Y .N

TURN ON OPR
ERR LIGHT

TURN ON PROGRAM
ALARM AND STORE
ALARM CODE
(00115)

ARE OPTICS AVAIL-
ABLE?

.Y .N

TURN ON PROGRAM
ALARM AND STORE
ALARM CODE
(00117)

MONITOR PROGRAM
ALARM LIGHT: IF THE
PROGRAM ALARM LIGHT
COMES ON AT THIS
TIME, THE ASTRONAUT
SHOULD KEY V05N09E
TO CHECK THE ALARM
CODE:

00115-SET OPTICS
MODE SW-CMC
RSET

00117-OPTICS NOT
AVAILABLE.
THE PROCESS
CANNOT BE
DONE AT THIS
TIME

#50

#60

#70

#80

++
+05
+
+05
++
EDIT

#90

#140

•
•
•

ENABLE CMC POSITION-
ING OF OPTICS TO
DESIRED ANGLES

```

      .
      .
      .
      .
      .
      .
      .
      .
      .
    ...
    .
EXIT
```

#160

REV 02 PCR MIT 66
REV 03 PCR MIT 66
REV 04 PCR 206
05 EDITORIAL